

Visualisation in Multidimensional Space: case of Robust Design

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Summary

1. Basic formulation of **Robust Design**
2. Application in **2D airfoil robust design**
optimisation
3. Visualisation in Multi-D: Self Organising
Maps

Robust Design Concept

In many real industrial optimisation, the input design parameters are **not fix**:

Uncertainties could characterise some geometric entities (lengths, relative positions, angles, etc.);

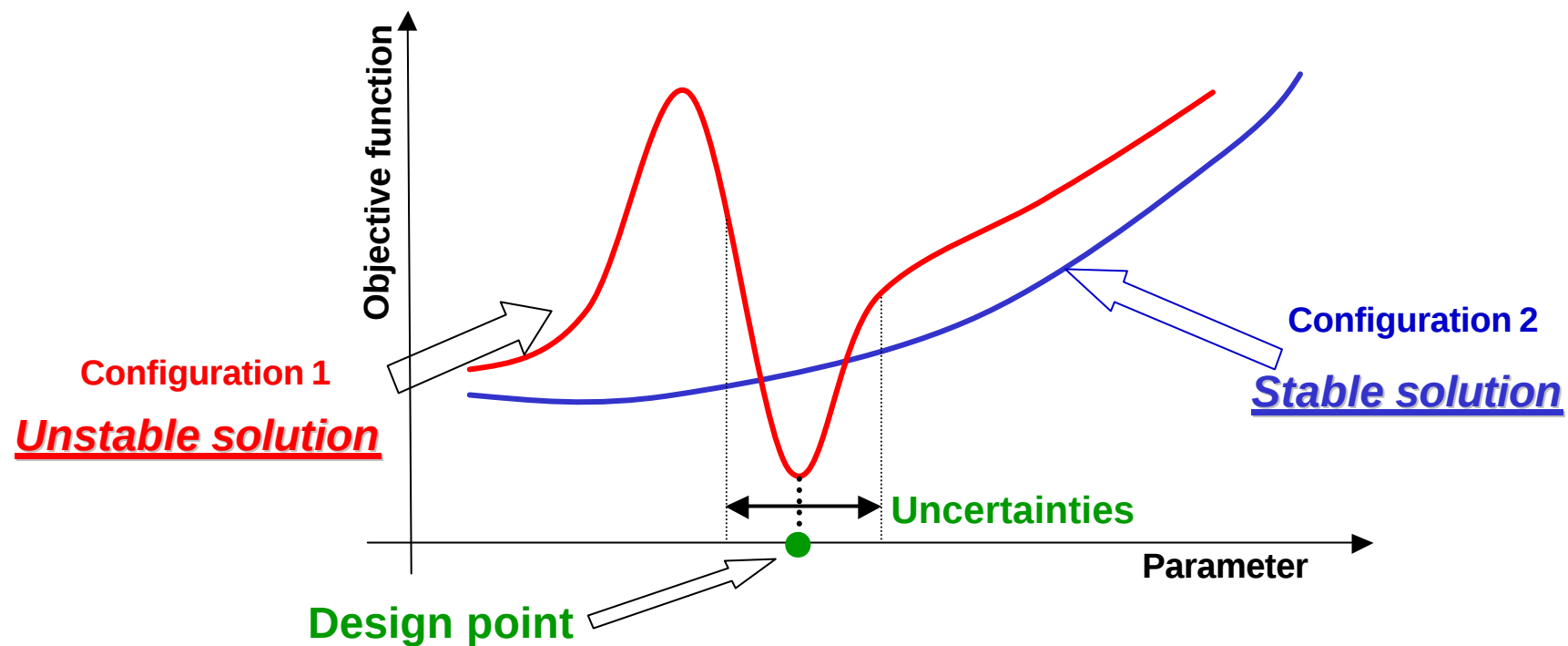
The operative conditions are not fixed cause of **fluctuations**

Turbomachines: the mass flow rate, inlet pressure, etc.

Aeronautic: flight speed, angle of attach, etc.

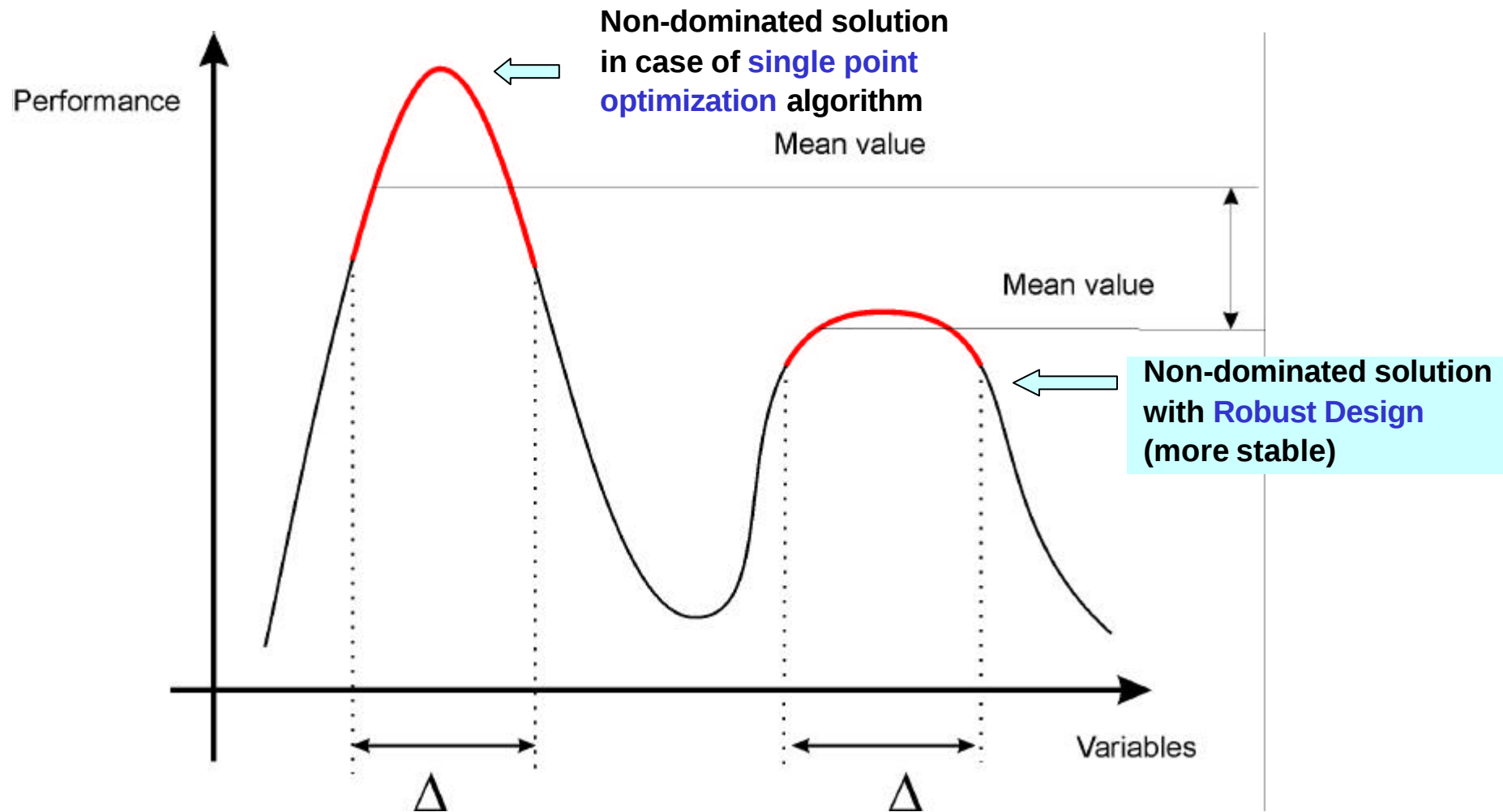
Robust Design Concept

- When there is the presence of **fluctuations** of the operating conditions it is important to define the **stability** of one solution;
- Traditional optimization techniques tend to “**over-optimize**”, producing solutions that perform well at the design point but have **poor off-design characteristics**.



ROBUST DESIGN

- What happens when we optimise a function in which the input design parameters are defined by the **mean** value (X_m) and the **deviation** ???



ROBUST DESIGN

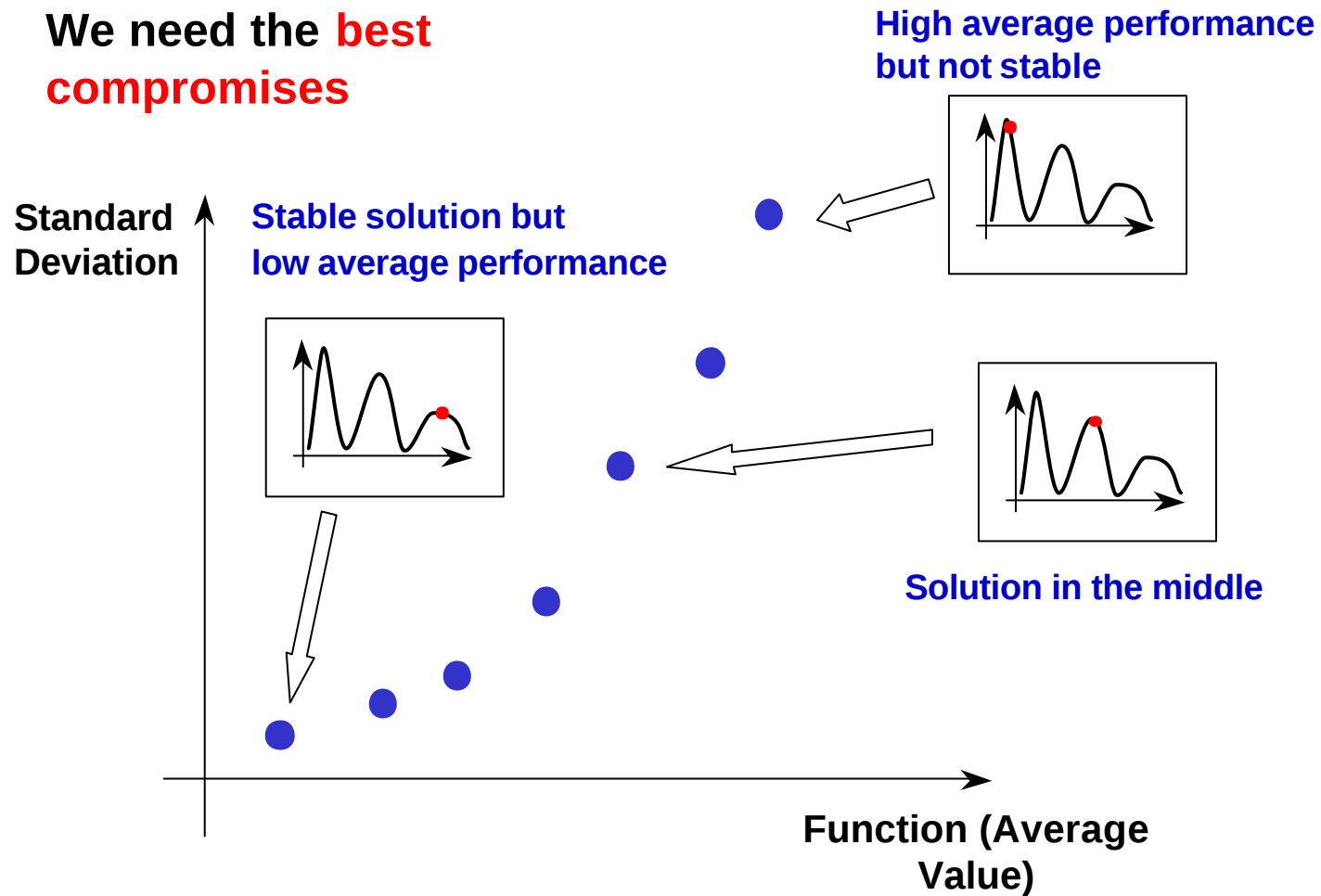
So when there is the presence of fluctuations a Multi Objective Approach is needed

- **Maximise** the mean value of the function
(performance)
- **Minimise** the variance of the function
(stability)

ROBUST DESIGN

What do we need?

We need the **best compromises**



ROBUST DESIGN

Mathematic formulation of the objective functions

$$\max f(x_i, q_i) \quad i = 1, \dots, k, \dots, n$$

$p(q_k)$ probability distribution of uncertainties q_k

with **Robust Design Theory** becomes:

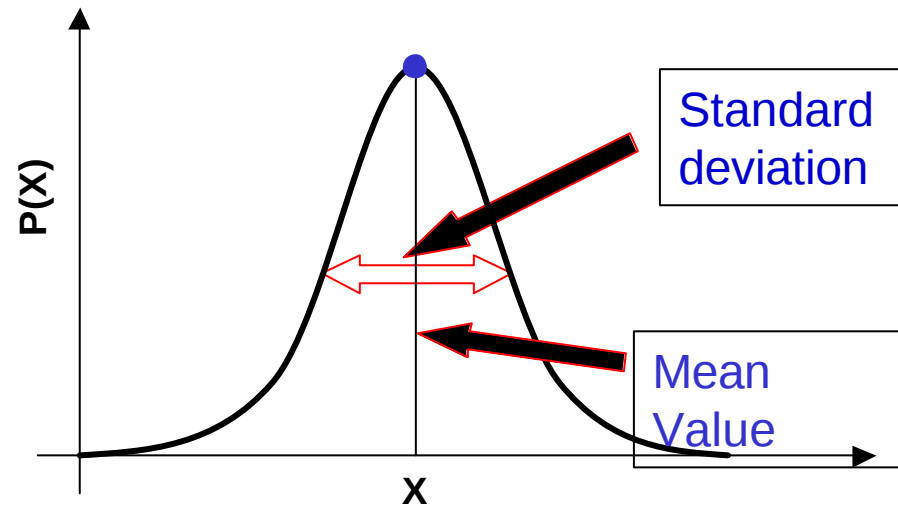
$$\max E(f_i) = \int f_i(x, u) p(u) du \quad \text{Mean value}$$

$$\min \sigma^2(f_i) = \int [f_i(x, u) - E(f_i)]^2 p(u) du \quad \text{Variance}$$

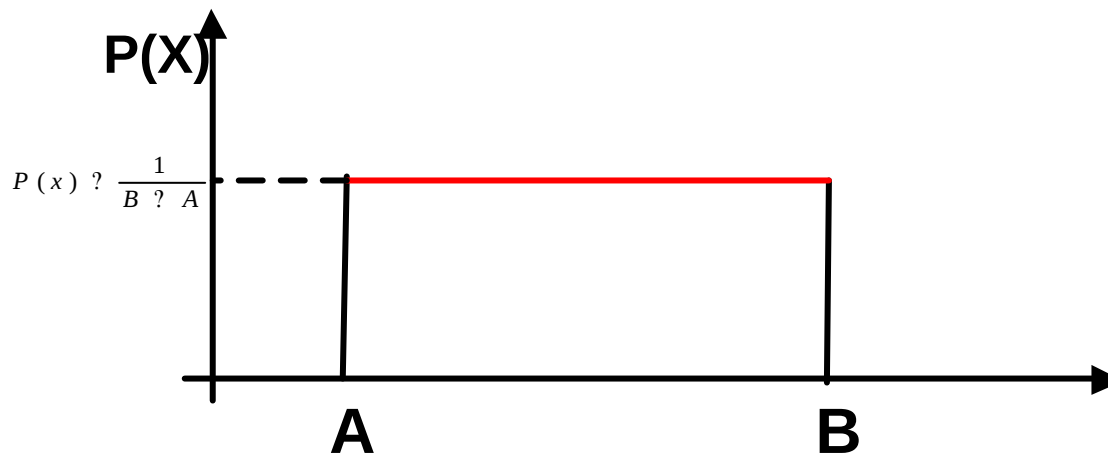
ROBUST DESIGN

It is possible to use different probability density functions:

Gaussian



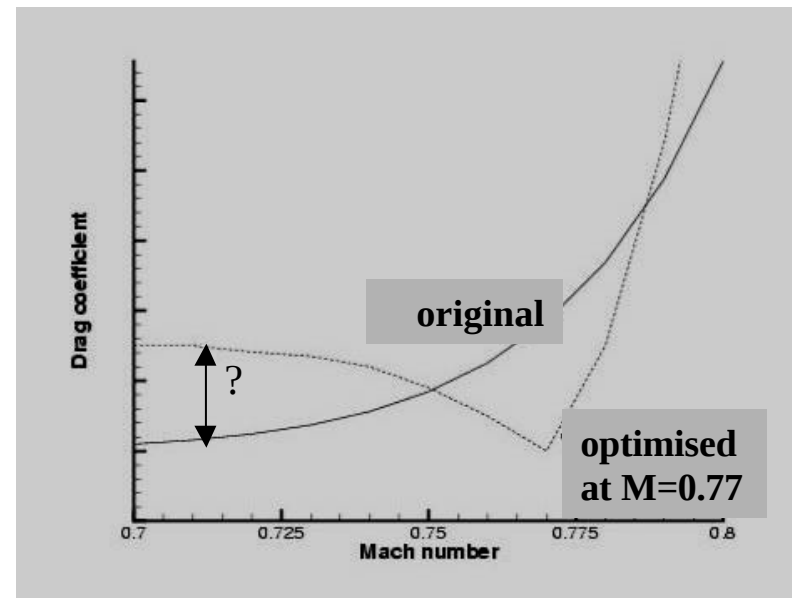
Uniform



Application in robust design airfoil optimization

- It is possible to illustrate the concept of **Robust Design** considering a 2D airfoil shape optimization problem in **transonic field**.

It has been observed (Hicks and Vanderplaats, 1977) that minimizing drag at a **single design point** causes reduction of performances (?) at **nearby off-design points**.



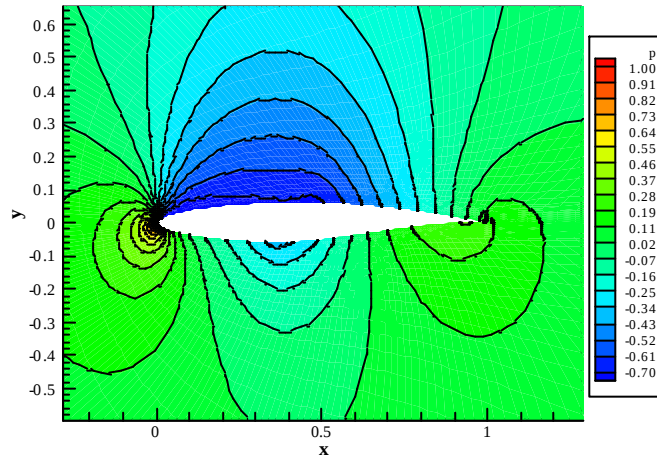
Thus, it is necessary to optimise drag with (two) input parameters given by **mean** values and **deviation**

$$\text{MACH} = 0.73 \pm 0.05 \quad ? = 2^\circ \pm 0.5$$

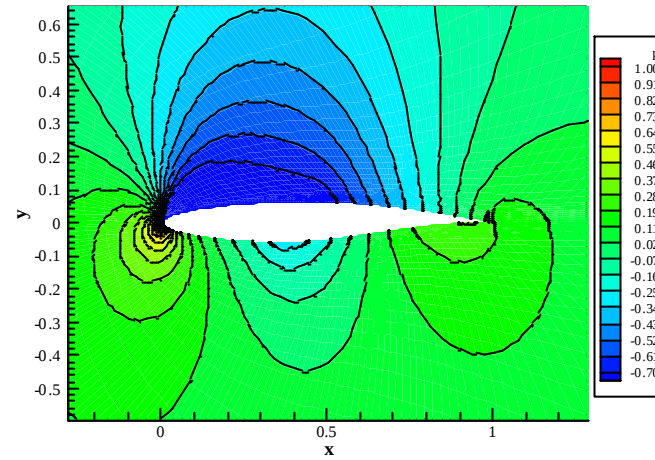
Uniform density function

Mach Field

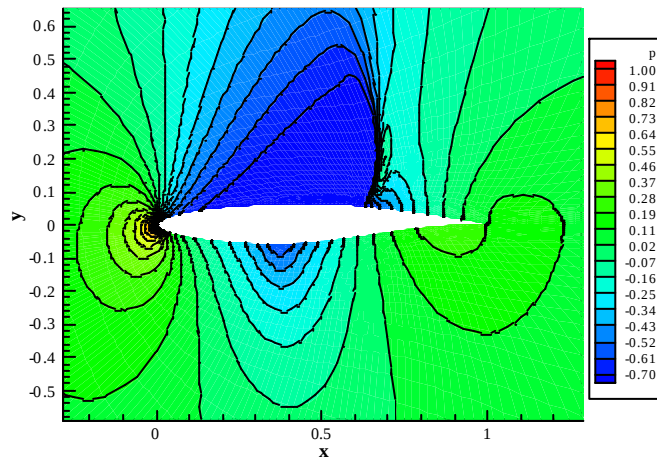
Mach 0.68, Alpha 1.5



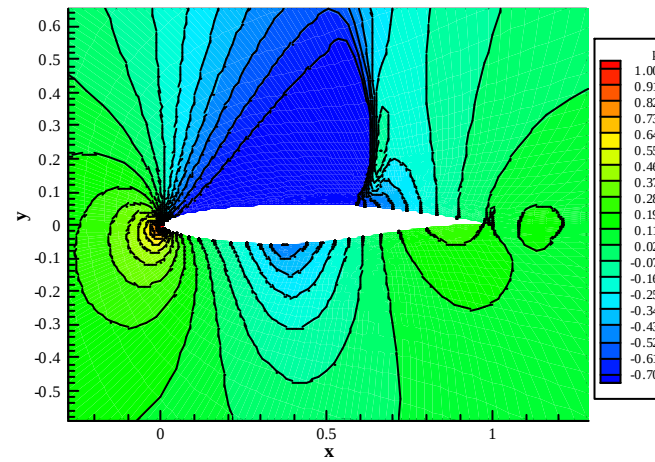
Mach 0.68, Alpha 2.5



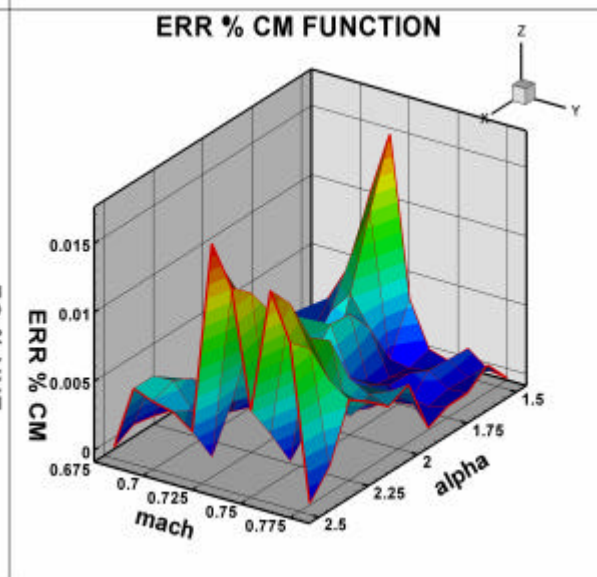
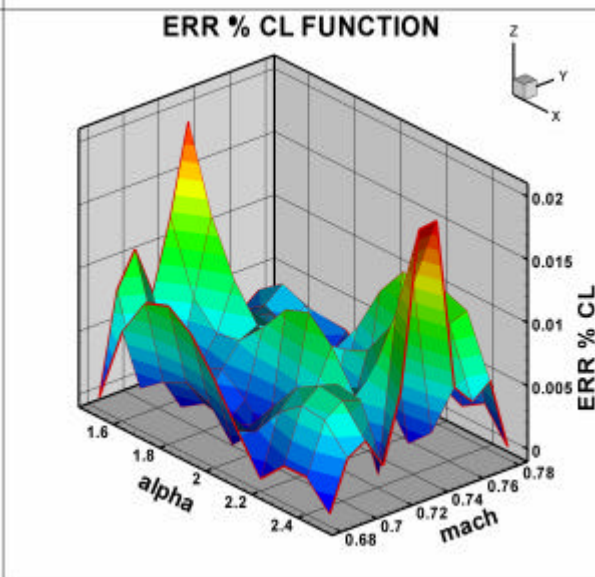
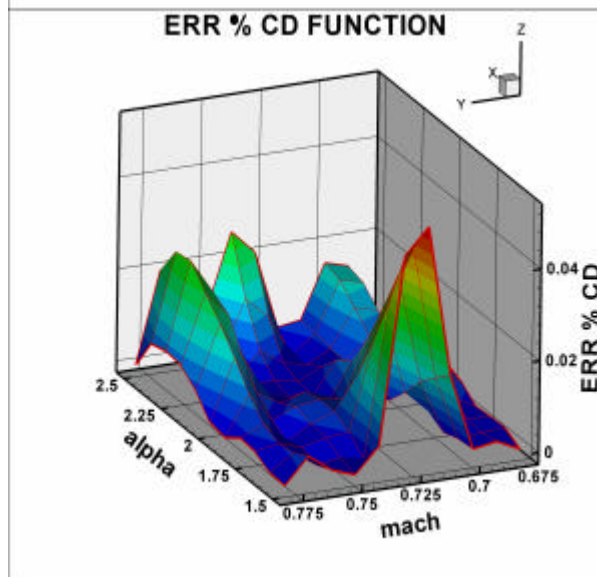
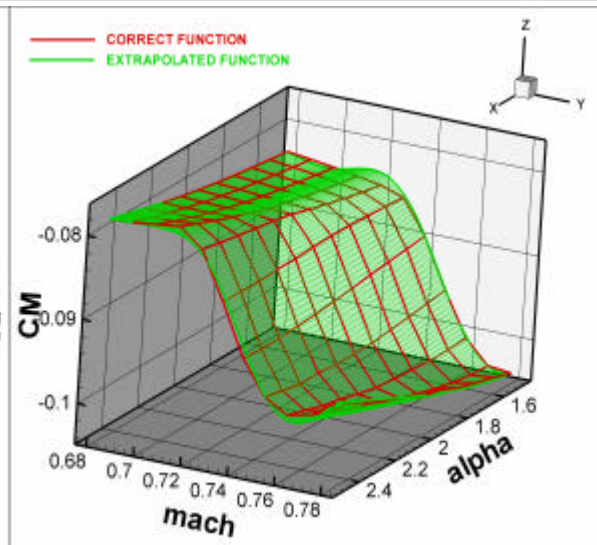
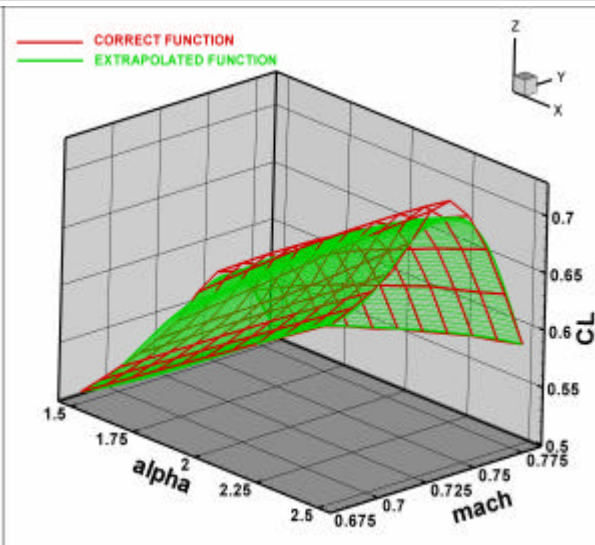
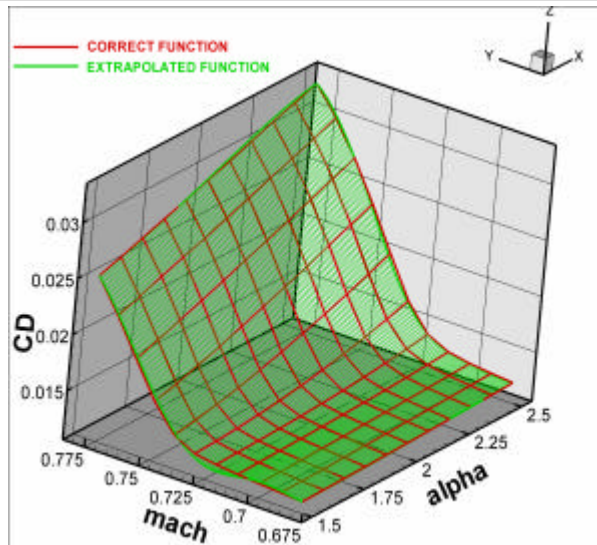
Mach 0.78, Alpha 1.5



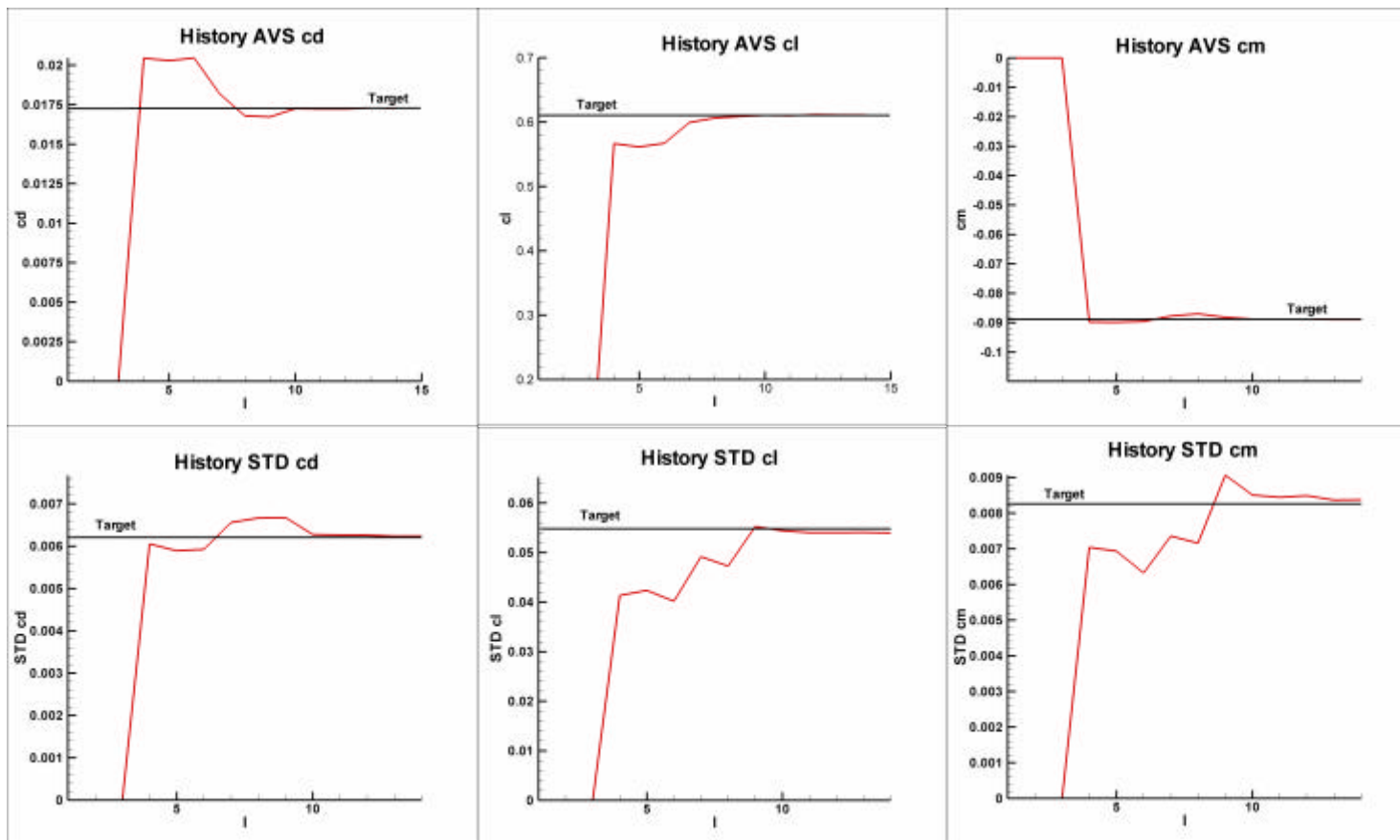
Mach 0.78, Alpha 2.5



REAL SURFACES C_l , C_d , C_m



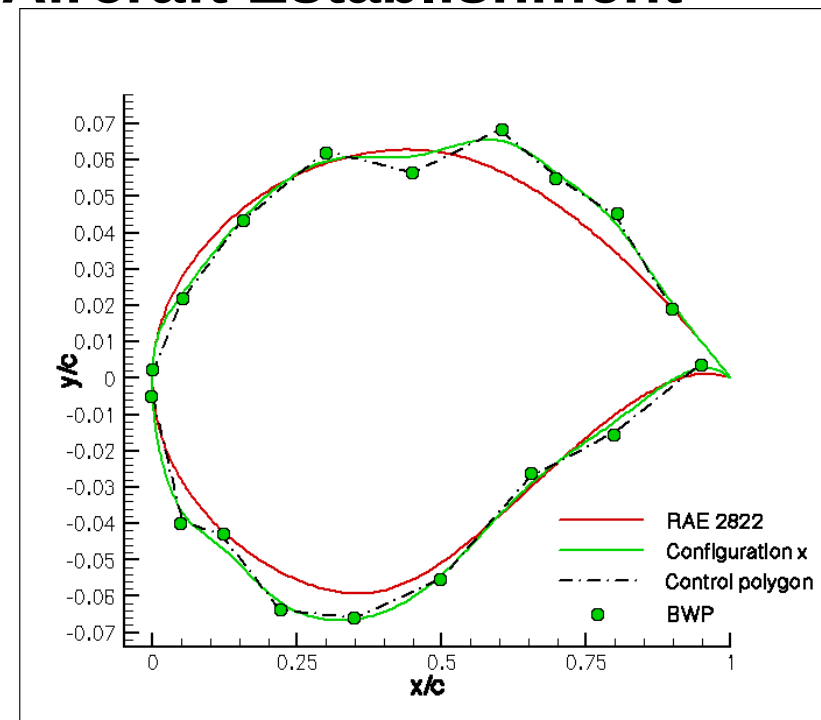
Convergence



10 iterations < 5% error

Parameterization and baseline configuration

- In the optimization process the achievable configurations have been determined by modifying an baseline configuration, the supercritical airfoil RAE 2822 designed by the Royal Aircraft Establishment
- Parameterisation of the airfoil by means 18 design variables (Bezièr weighting points).



TEST CASE: RAE2822

This problem of **Robust Design** can be expressed as:

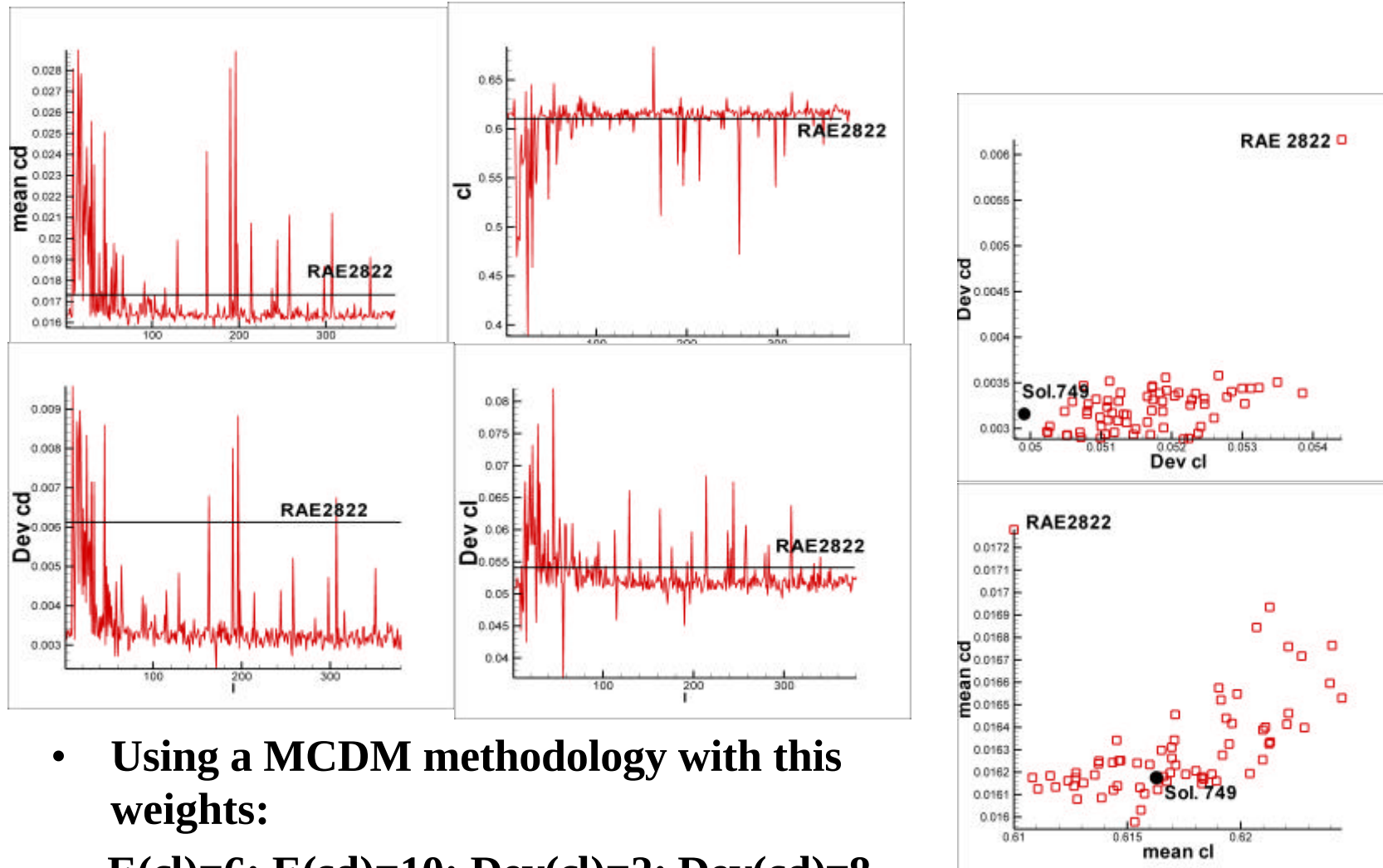
$$\begin{array}{ll}
 \max & E(c_l) \\
 \min & \sigma^2(c_l) \\
 \min & E(c_d) \\
 \min & \sigma^2(c_d)
 \end{array}
 \left. \vphantom{\begin{array}{l} \max \\ \min \\ \min \\ \min \end{array}} \right\} \begin{array}{l} \text{Lift} \\ \text{Drag} \end{array}$$

with **constraints**:

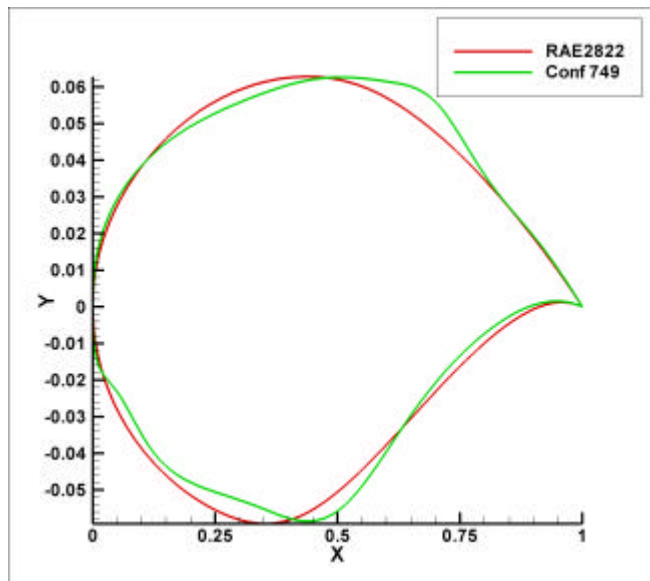
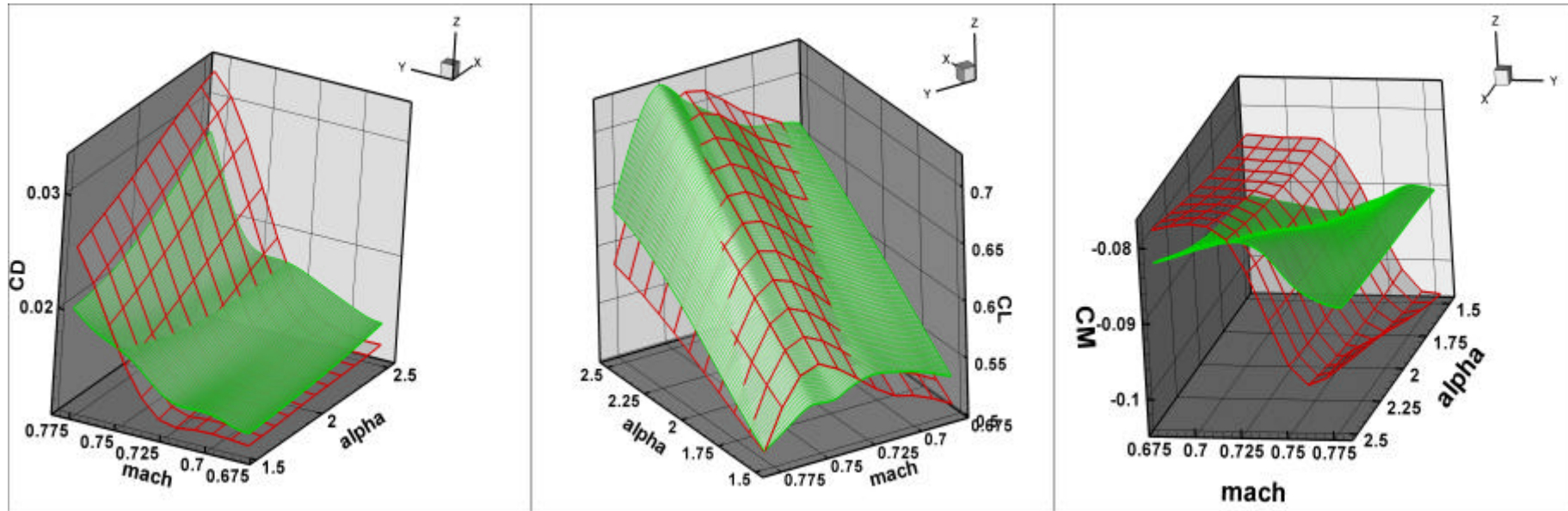
$$\begin{cases} E(c_l) \leq E(c_l)^* \\ E(c_d) \leq E(c_d)^* \\ E(c_m) \leq E(c_m)^* \end{cases} \quad \begin{cases} \sigma^2(c_l) \leq \sigma^2(c_l)^* \\ \sigma^2(c_d) \leq \sigma^2(c_d)^* \\ \sigma^2(c_m) \leq \sigma^2(c_m)^* \end{cases}$$

where $E(c_l)^*, E(c_d)^*, E(c_m)^*, \sigma^2(c_l)^*, \sigma^2(c_d)^*, \sigma^2(c_m)^*$ are respectively the mean values and the variances of the original airfoil (RAE2822).

TEST CASE: RAE2822



RAE2822 (Results)



Obiettivi	RAE2822	Sol. 749
$E(C_l)$	0,61	0,616
$E(C_d)$	0,017	0,01616
$Dev(C_l)$	0,05441	0.04976
$Dev(C_d)$	0,06164	0,03434

Visualization in Multi-D (SOM)

Clustering of data ? **Self-Organizing Maps**

SOM is a two-dimensional array of neurons

$M = \{m_1, \dots, m_{p \times q}\}$ m same dimension as the input vectors
(objectives)

Best-matching unit:

$$\|x - m_c\| = \min_i \|x - m_i\|$$

Algorithm SOM – Basic Idea

- T. Kohonen* in the '80
- The basic idea is the capacity of the brain to organize in topographical way the data; every stimulus is proceeded in specific zones.
- The numerical algorithms which use this basic idea are called *self-organizing*; the Self Organizing Maps are the main example.
- The SOM performs the association between the output and ordered units (in 2D space); so it is possible to have an ordered view of the data.

Algorithm SOM

1) $\mathbf{X}^T = (X_1, X_2 \dots X_d)$, data to be represented.

Build a grid with i nodes ($i=1, \dots, k$) in 2D. Every node get a d -dimensional vector $\mathbf{W}^{(i)T} = (W_1^{(i)} \ W_2^{(i)} \dots W_d^{(i)})$, $\mathbf{W}^{(i)} = \mathbf{W}^{(i)}(t)$ (*codebook*), changes iteratively.

2) Initialization: random values $\mathbf{W}^{(i)}(0)$ for the nodes $i=1 \dots K$.

Initial parameters

neighborhood function

$\sigma(t) h(|r_i - r_c| / \sigma(t), t)$

3) Selection of one $\mathbf{X}^T$

Algorithm SOM

3.1) Distance $d(\mathbf{X}, \mathbf{W}^{(i)})$, find $\mathbf{W}^{(c)}$ similar to (closer to) $\mathbf{X}$

3.2) Update of every component values in the $O(r_c)$

3.3) Decrease $h_o(t)$ and $\sigma(t)$.

3.4) End of the iterations

4) Quantisation errors
(quality of the maps)

$$E_q = \sum_{c=1}^m \sum_{i=1}^n \|\mathbf{x}_i - \mathbf{w}_c\|^2$$

Utility of SOM

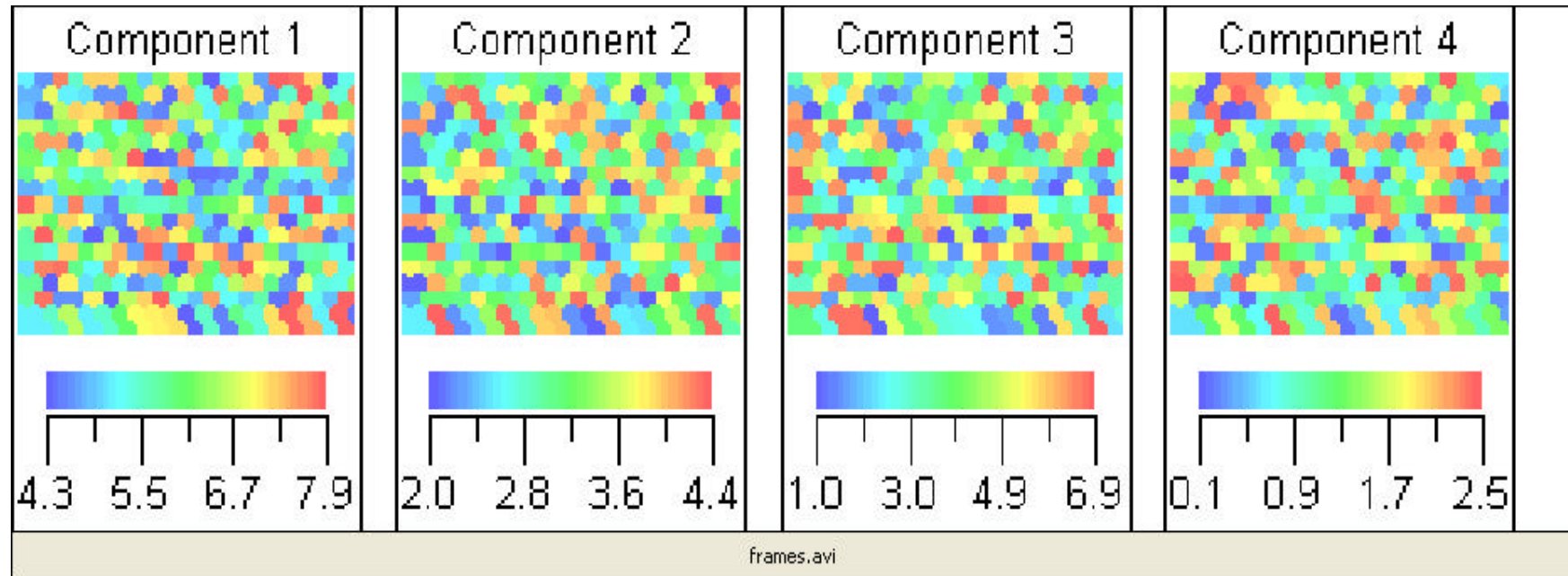
-  Visualisation of data n-dimensional
-  Exploration of data
-  Classification

Utility of SOM- Iris - Data

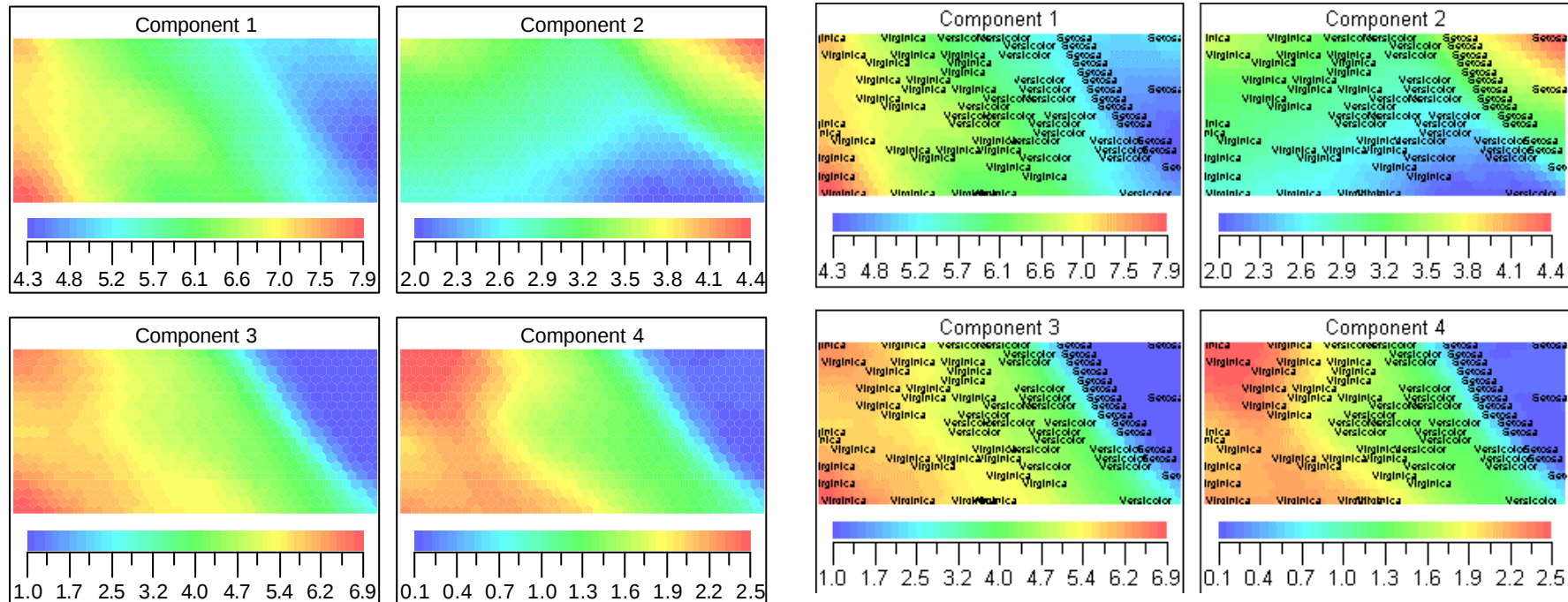
SepalL	SepalW	PetalL	PetalW	Type
4.6	3.6	1.0	0.2	Setosa
5.1	3.3	1.7	0.5	Setosa
4.8	3.4	1.9	0.2	Setosa
5.0	3.0	1.6	0.2	Setosa
5.0	3.4	1.6	0.4	Setosa
6.5	2.8	4.6	1.5	Versicolor
5.7	2.8	4.5	1.3	Versicolor
6.3	3.3	4.7	1.6	Versicolor
4.9	2.4	3.3	1.0	Versicolor
6.6	2.9	4.6	1.3	Versicolor
7.6	3.0	6.6	2.1	Virginica
4.9	2.5	4.5	1.7	Virginica
7.3	2.9	6.3	1.8	Virginica
6.7	2.5	5.8	1.8	Virginica
7.2	3.6	6.1	2.5	Virginica
6.5	3.2	5.1	2.0	Virginica

150 data, 4 variables
3 typology of Iris

Algorithm SOM - Dynamic



IRIS EXAMPLE



Local correlation
non linear,

Every data is a node

Clustering

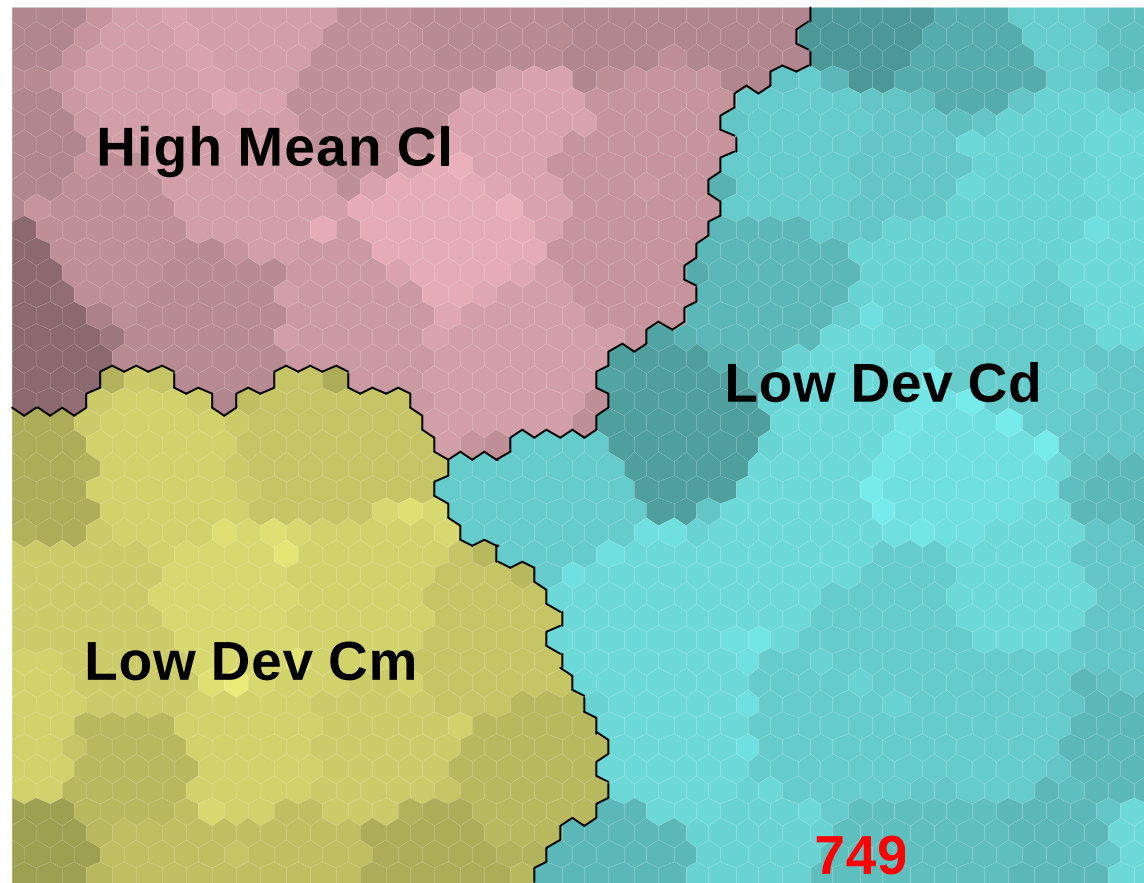
U-Matrix



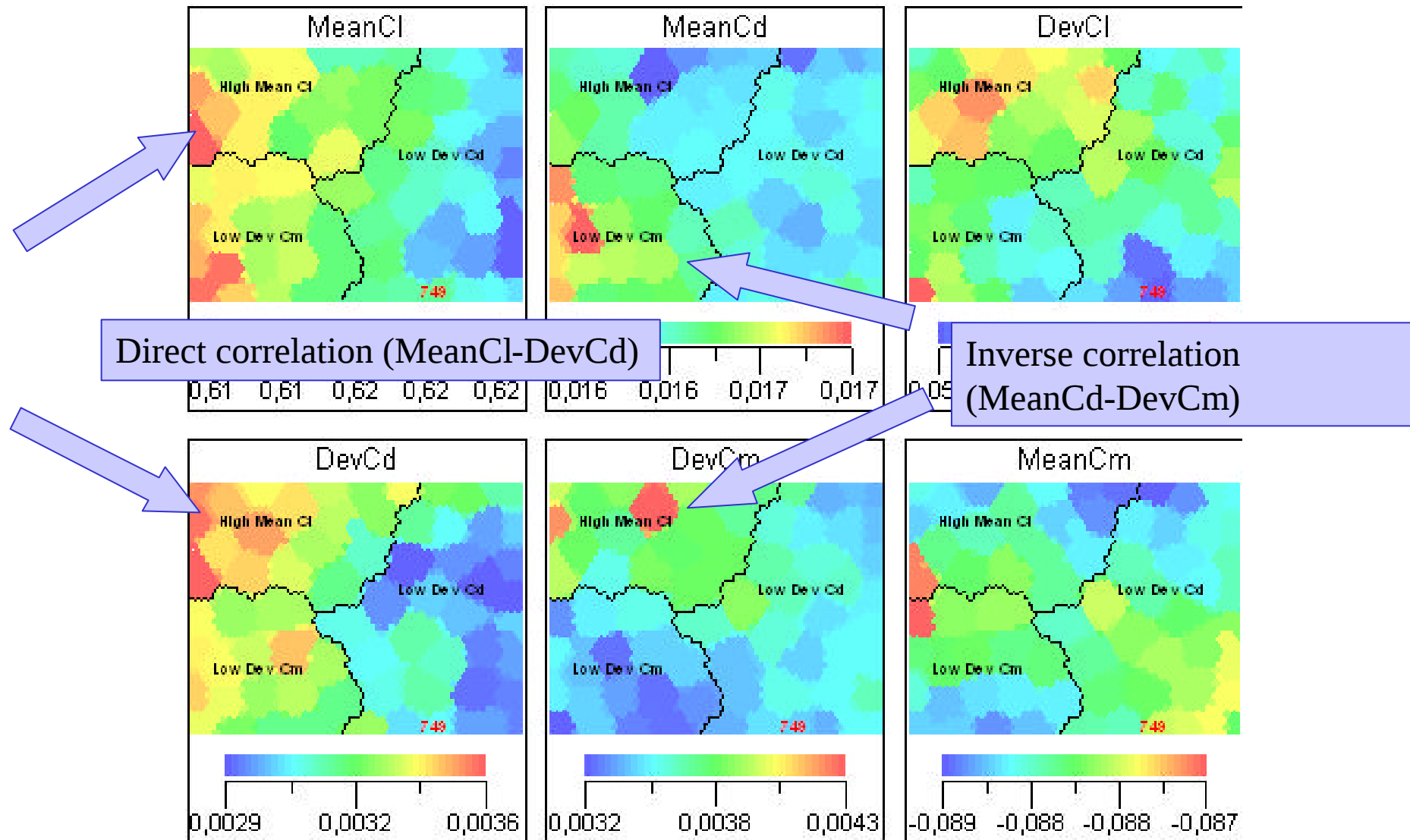
Clustering no supervised: with the U-matrix, based on the distance Between the data, it is possible to rappresent the *cluster* (collection of similar data)

Visualization in Multi-D (SOM)

SOM with the data of the Pareto Front (previous optimization)

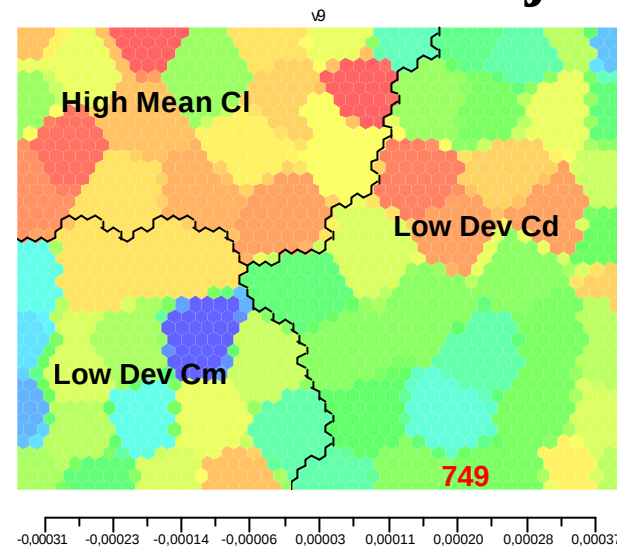
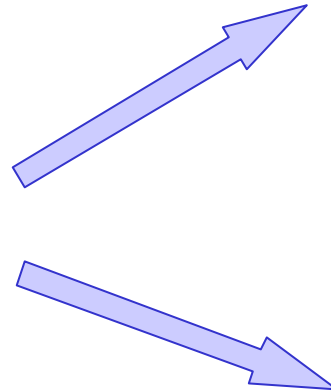
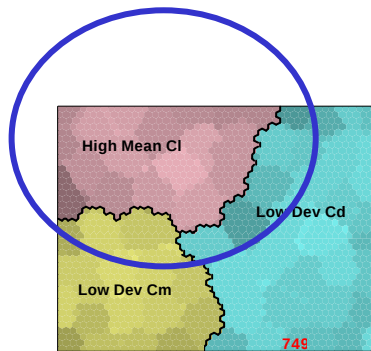


Visualization in Multi-D (SOM)

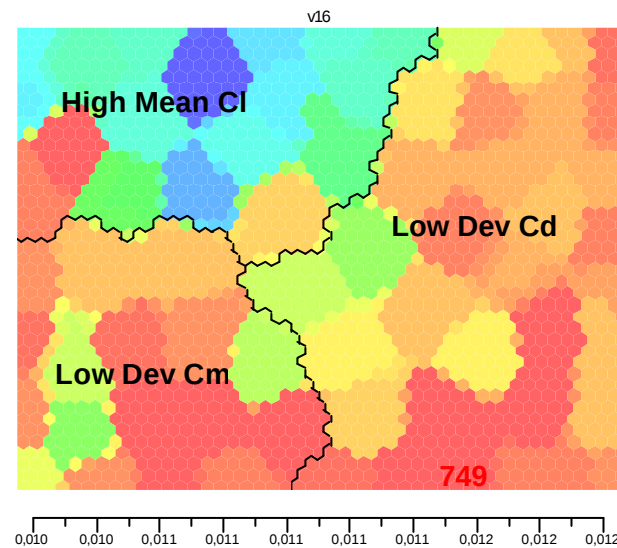


Visualization in Multi-D (SOM)

Easy visualization of the relations between objectives and variables

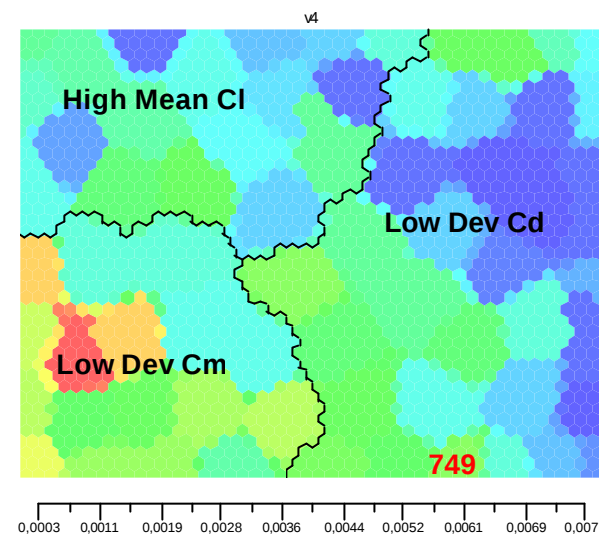
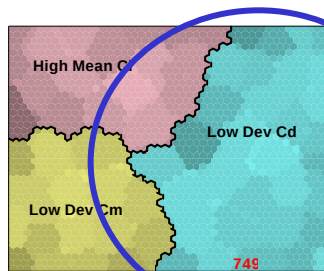


V9

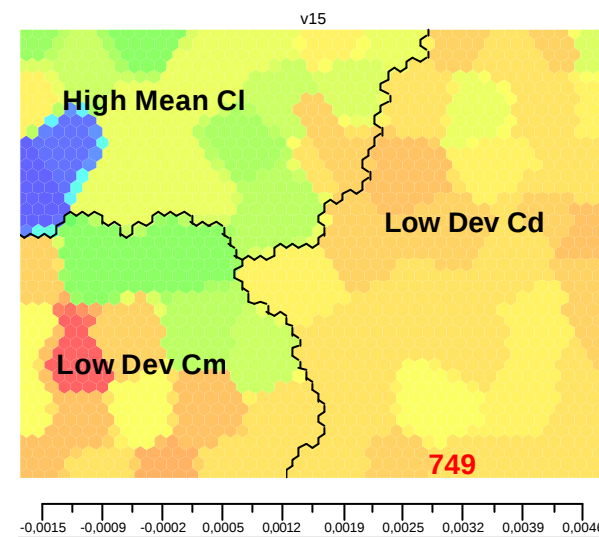


V16

Visualization in Multi-D (SOM)

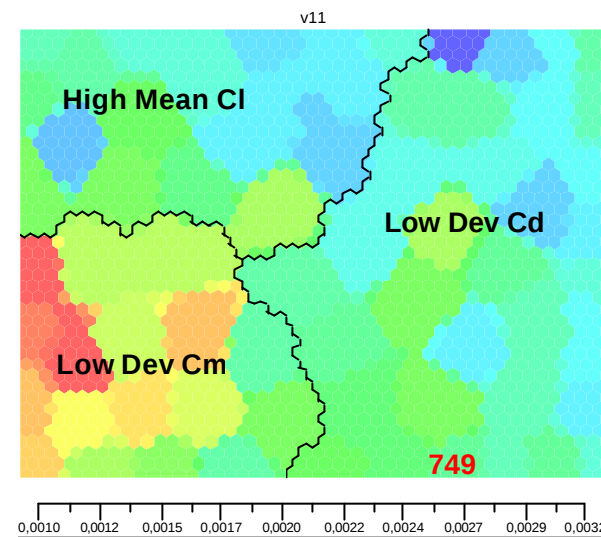
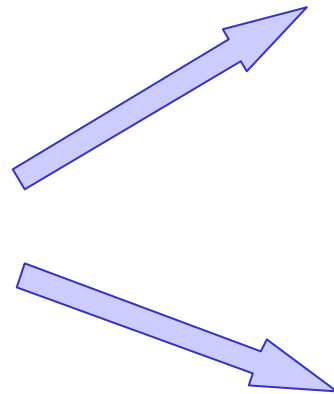
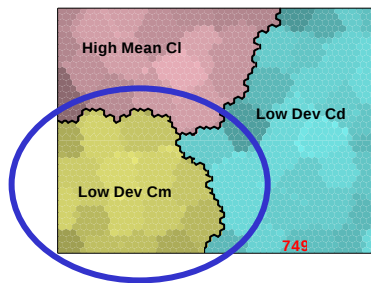


V4

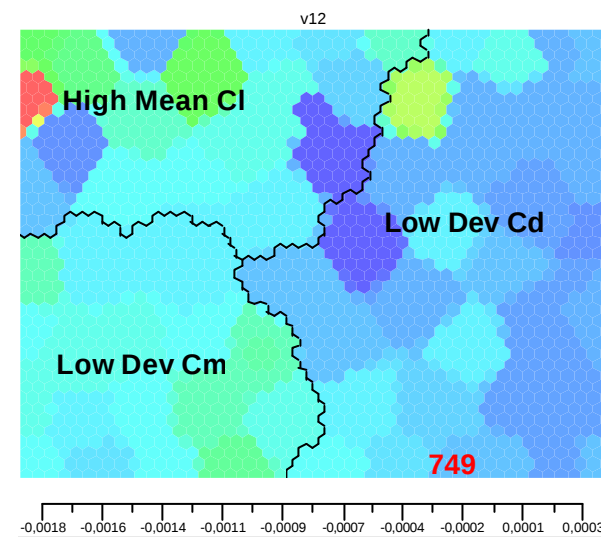


V15

Visualization in Multi-D (SOM)



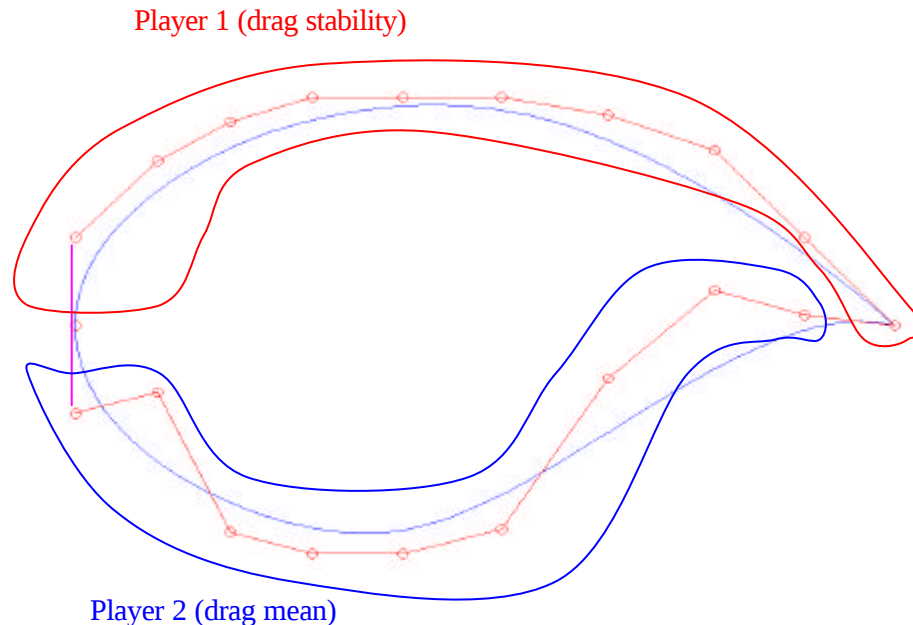
V11



V12

Game theory approach (Nash)

Objectives: **minimise** Mean(C_d) and Std (C_d)



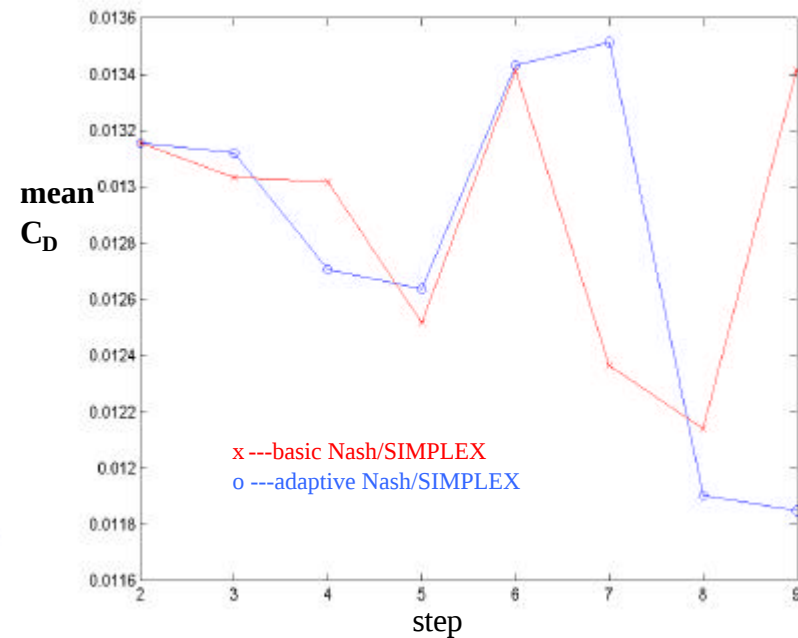
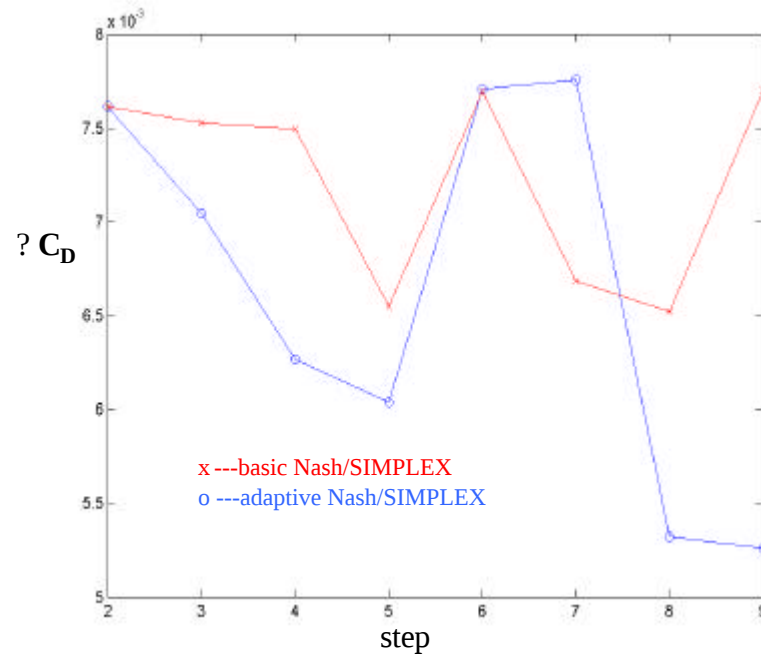
Initial variables space decomposition:

9 co-ordinates Bézier points upper side – standard deviation of drag

9 co-ordinates Bézier points lower side – mean value of drag

Game theory approach (Nash)

Results of optimisation: basic algorithm and adaptive one

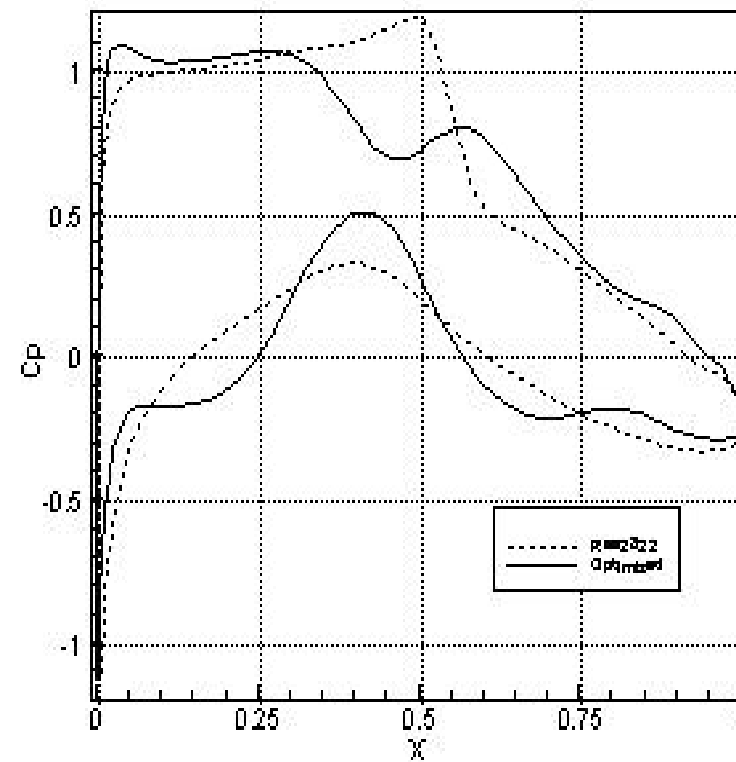
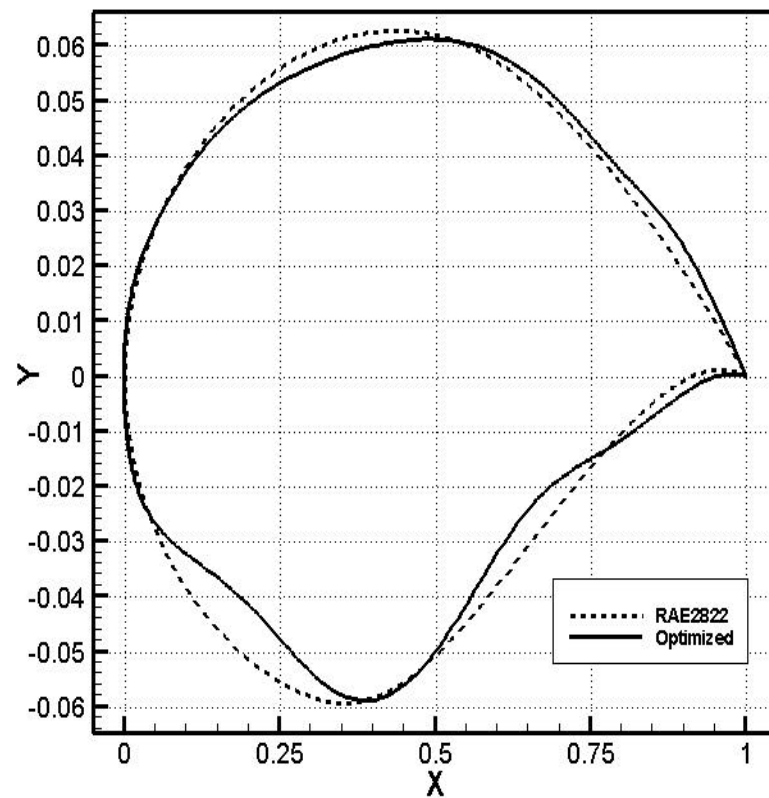


	MeanCd (obj)	? Cd (obj)	MeanCl (constr)	? Cl (constr)	MeanCm (constr)	? Cm (constr)
RAE2822	1.33e-2	7.6e-3	0.686	5.91e-2	-0.105	9.50e-3
BEST	1.18e-2	5.27e-3	0.691	5.83e-3	-0.102	6.45e-3

Total number of design: about **500**

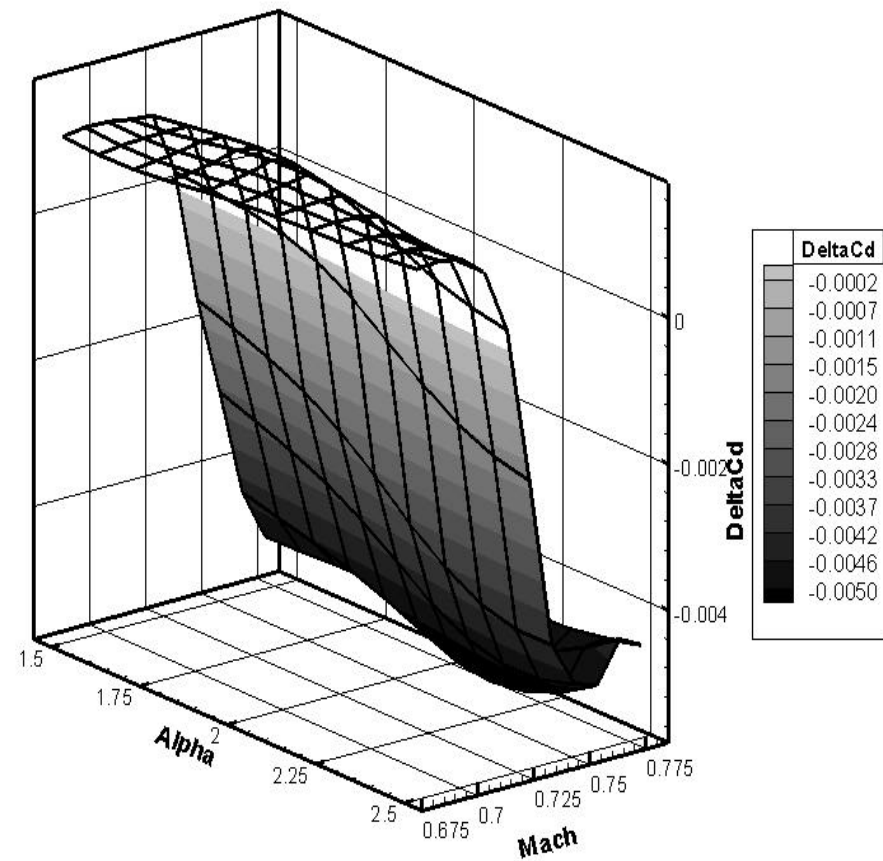
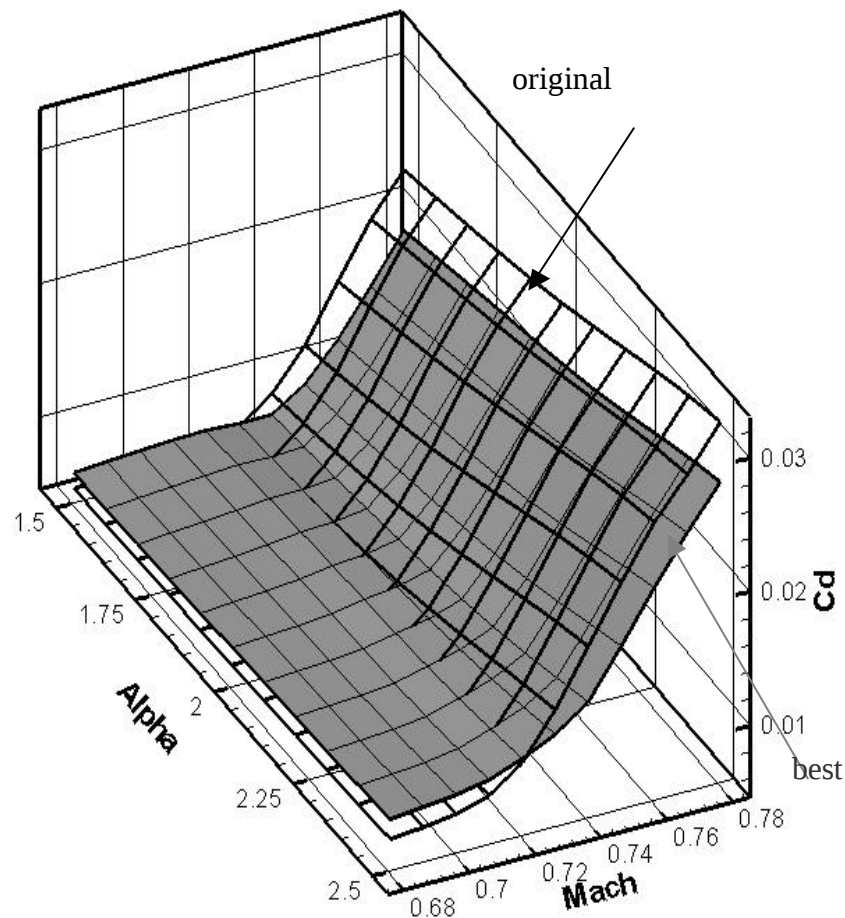
Game theory approach (Nash)

Comparison of original RAE with best configuration



Game theory approach (Nash)

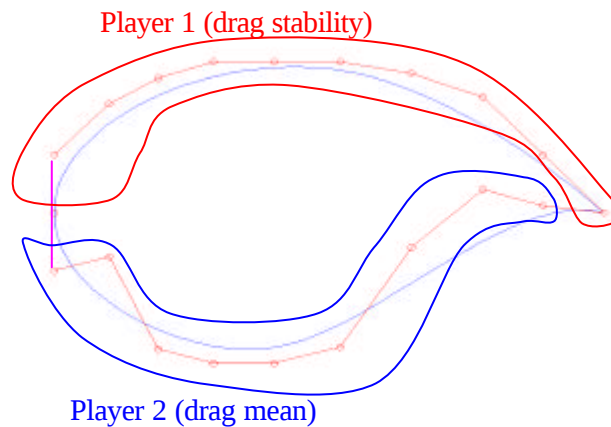
Drag surface of best configuration: difference with respect to RAE



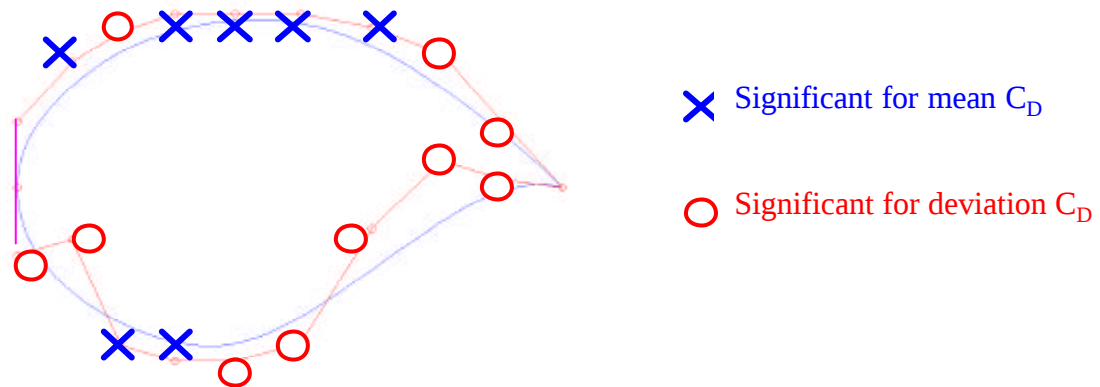
Game theory approach (Nash)

Final variables space decomposition found by adaptive Nash/SIMPLEX

Original guess



Final decomposition



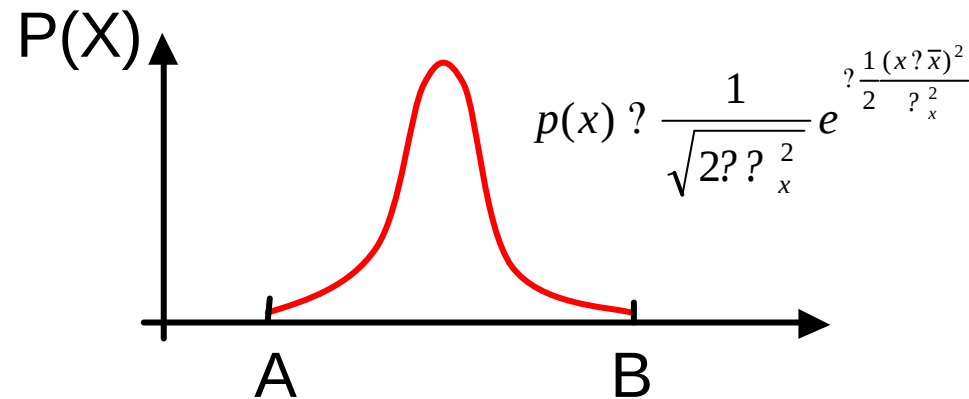
Multi Point with Gaussian probability distribution

2 design point:

SUBSONIC $\alpha = 9^\circ$ ($\Delta\alpha = 0.33^\circ$) Mach=0.12 ($\Delta\text{Mach} = 0.0066$)

TRANSONIC $\alpha = 2^\circ$ ($\Delta\alpha = 0.16^\circ$) Mach=0.73 ($\Delta\text{Mach} = 0.01$)

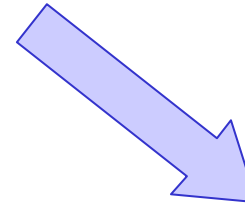
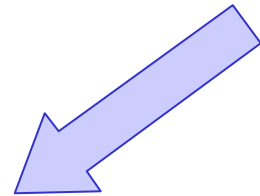
Gaussian probability distribution of uncertainties



Gaussian distribution

Multi Point with Gaussian probability distribution

4 Objectives



Subsonic

Max Mean Efficiency

Min Dev Efficiency

Transonic

Min Mean Cd

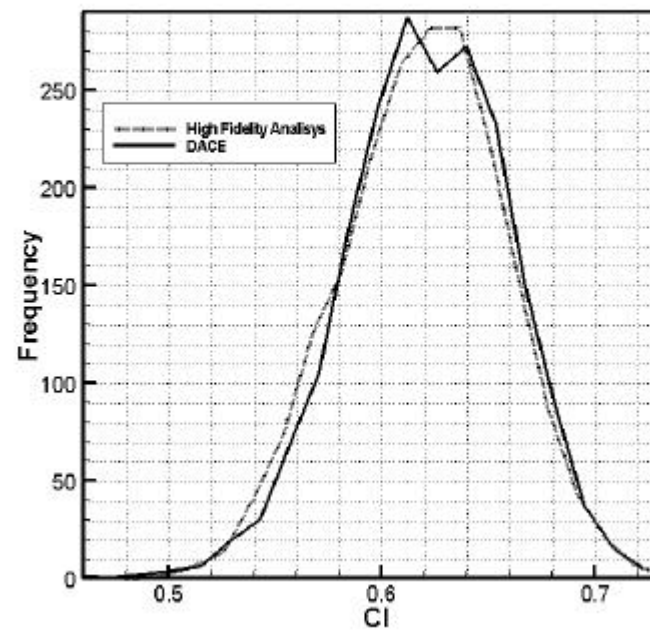
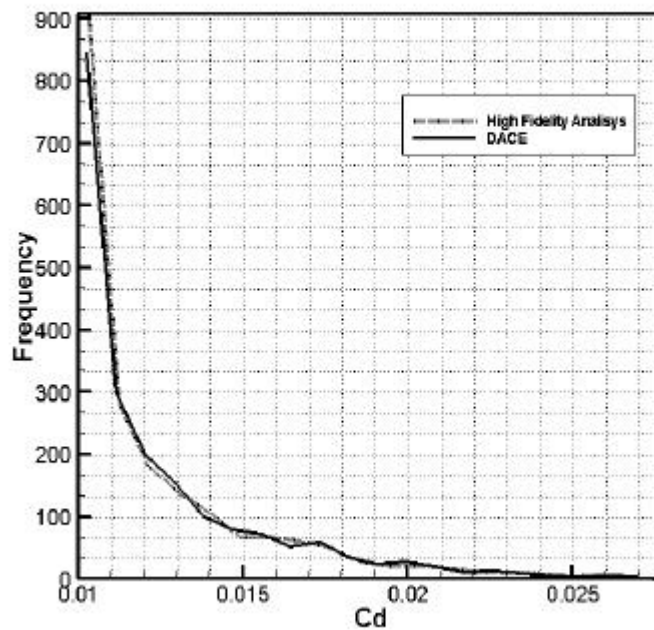
Min Dev Cd

Constraints performances better than the RAE (thickness too)

DACE

CFD code: Euler + Boundary layer (Drela code, Mses)

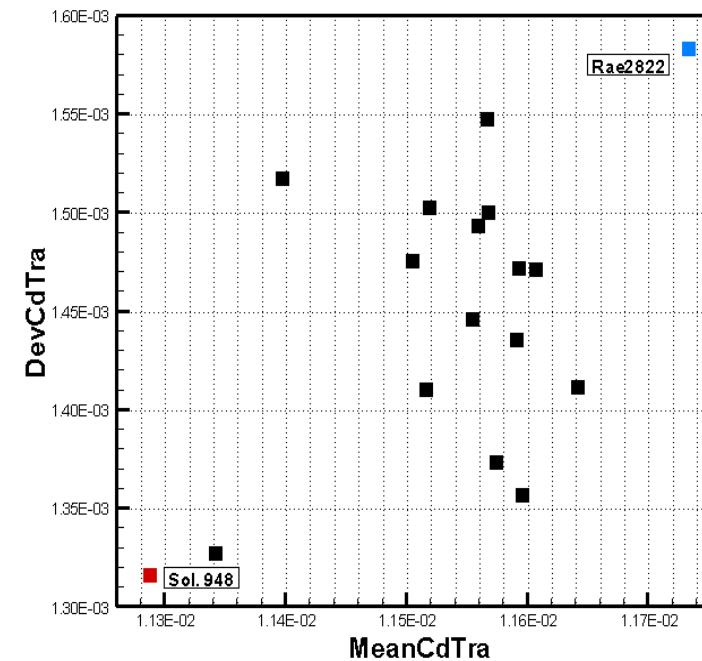
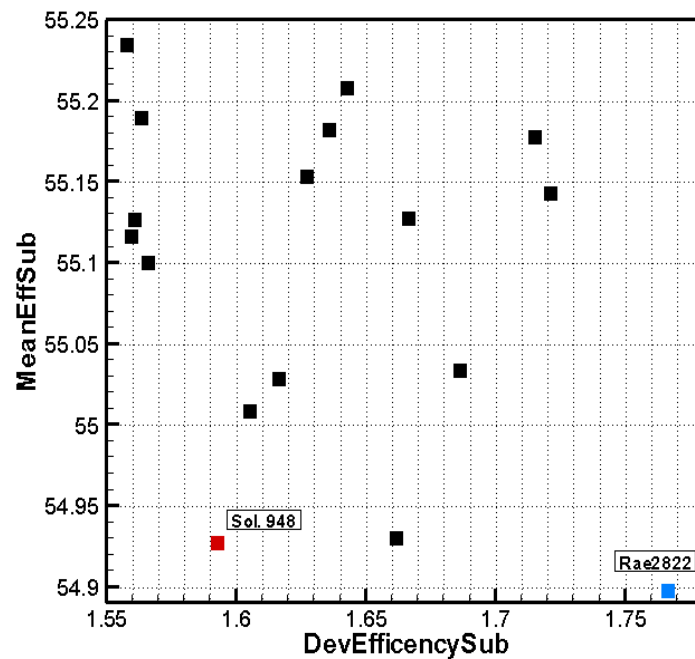
Very interesting results with Adaptive Dace Approach



Optimization Approach

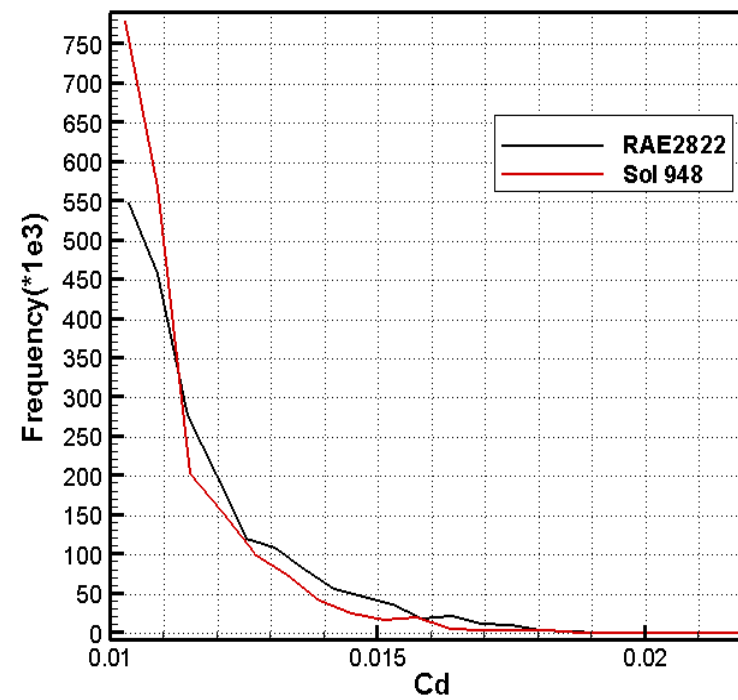
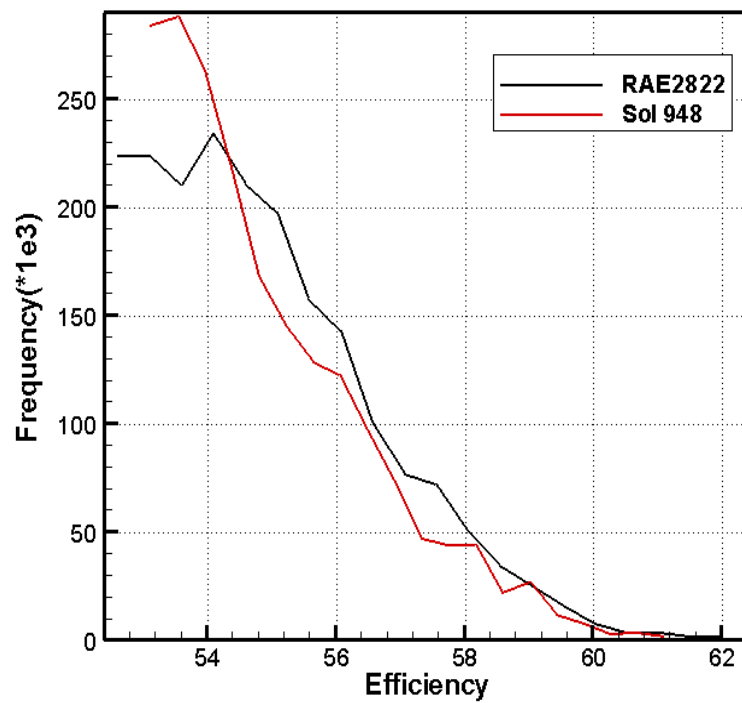
A **MOGA** has been used (60 ind, 16 gen.)

CODASID for determining the best in the Pareto Front
(MeanEff=10, DevEff=6, MeanCd=10, DevCd=8)

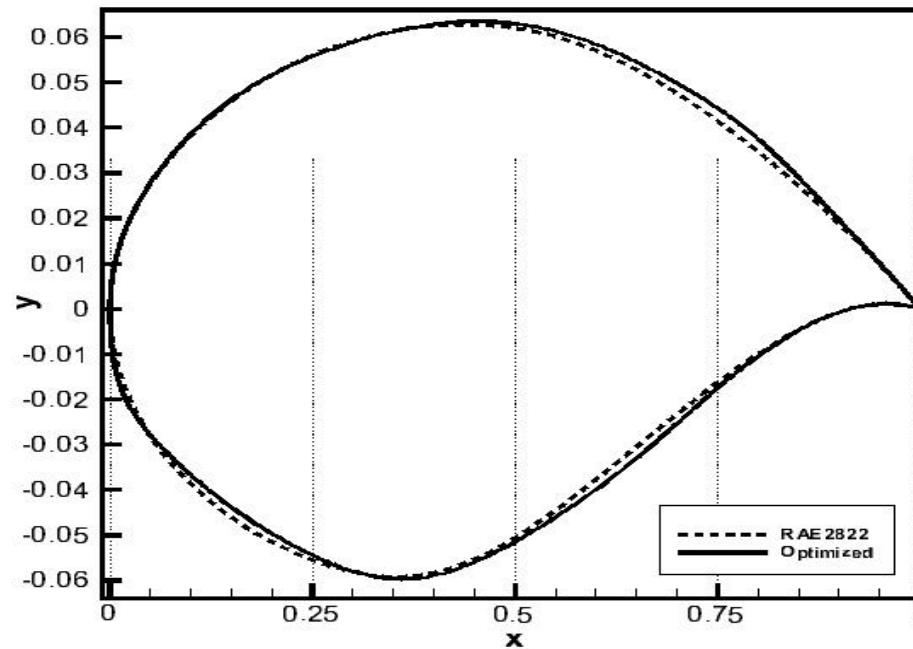


RESULTS

More importance to drag (Mean and Std)



AIRFOIL COMPARISON

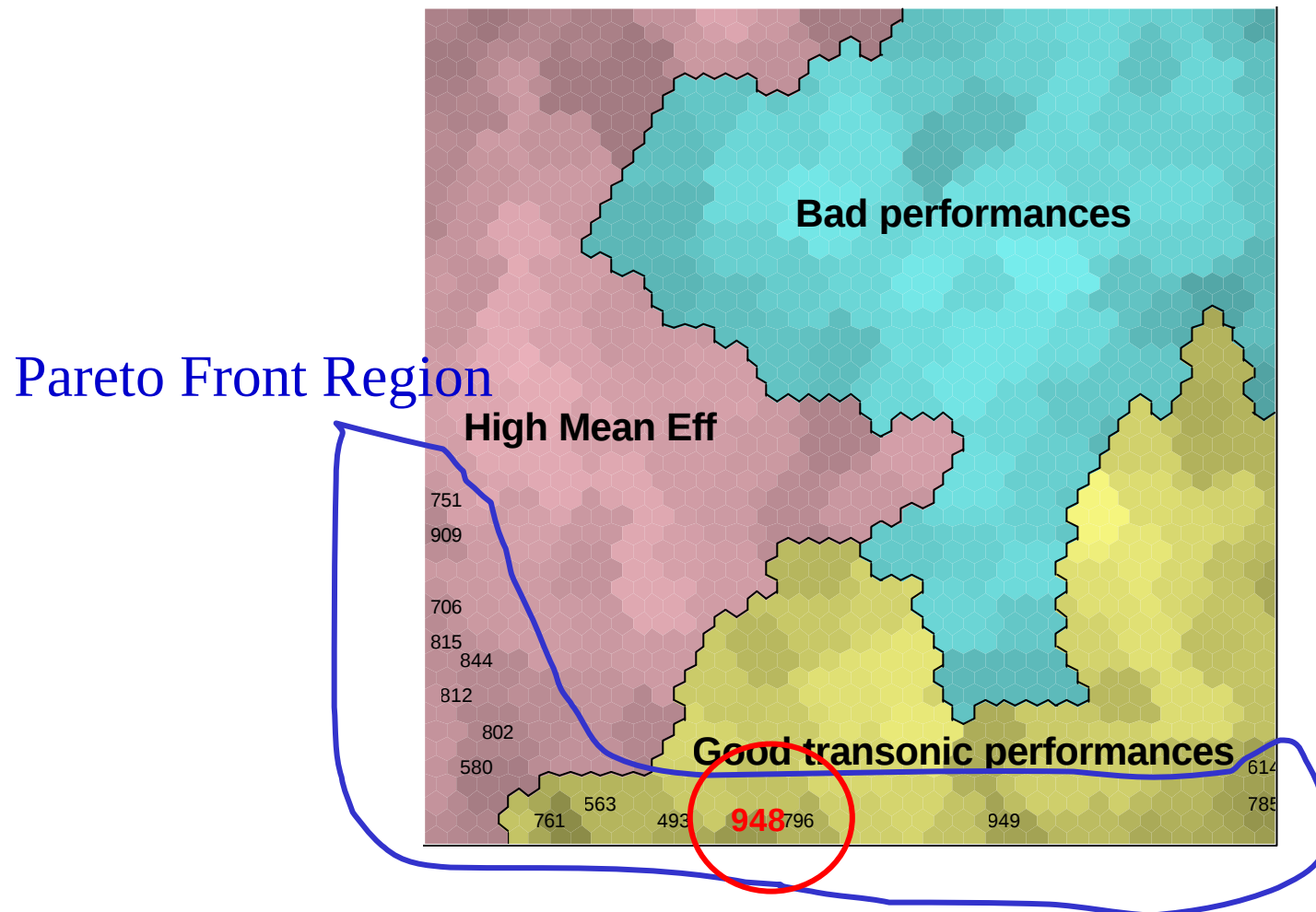


Objectives	Rae2822	Sol 948
E(Eff)	54,89	54,92
E(Cd)	1,173E-02	1,128E-02
Dev(Eff)	1,766	1,593
Dev(Cd)	1,583E-03	1,316E-03

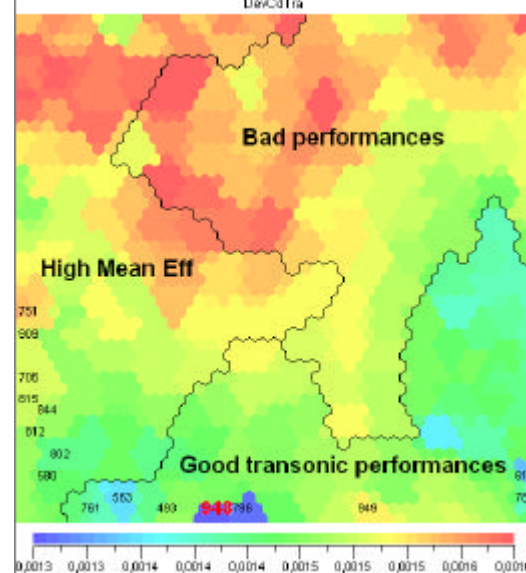
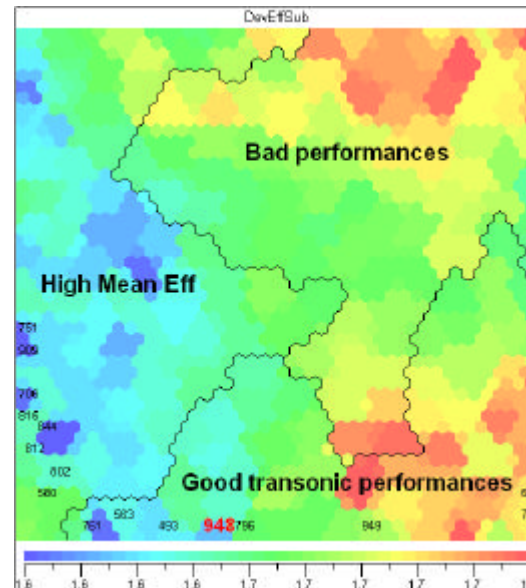
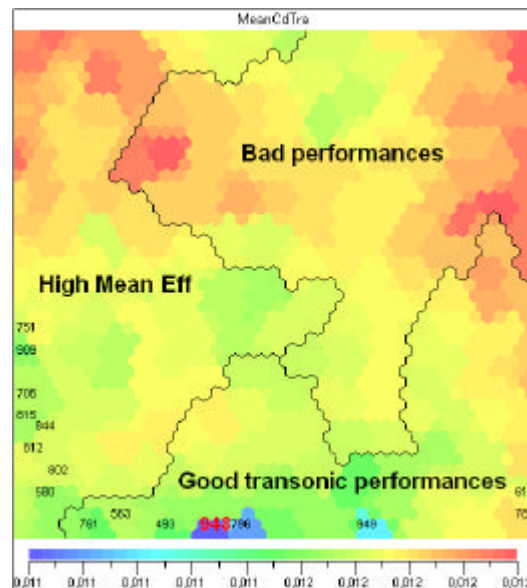
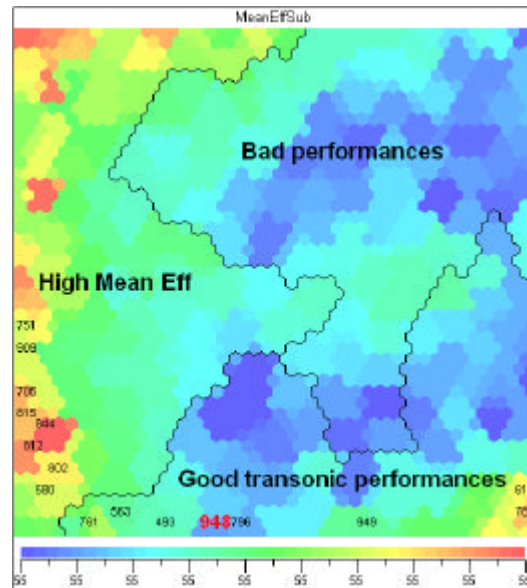
**Better results
for stability**

Visualization in Multi-D (SOM)

It is possible to represent all the feasible configuration with 3 clusters, **with the Pareto Front and MCDM best configuration**



Visualization in Multi-D (SOM)



The names of the clusters are easy to fix

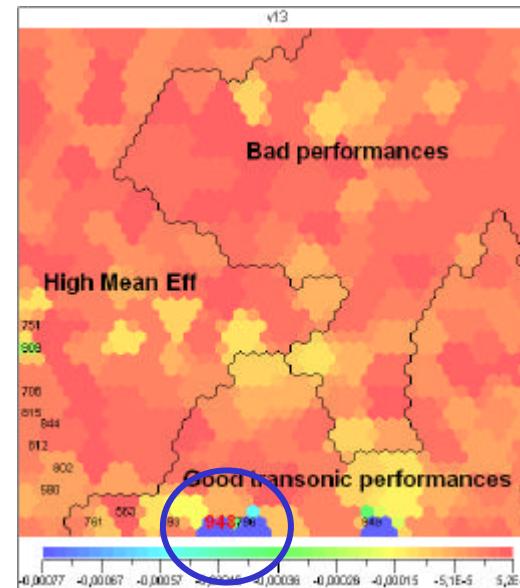
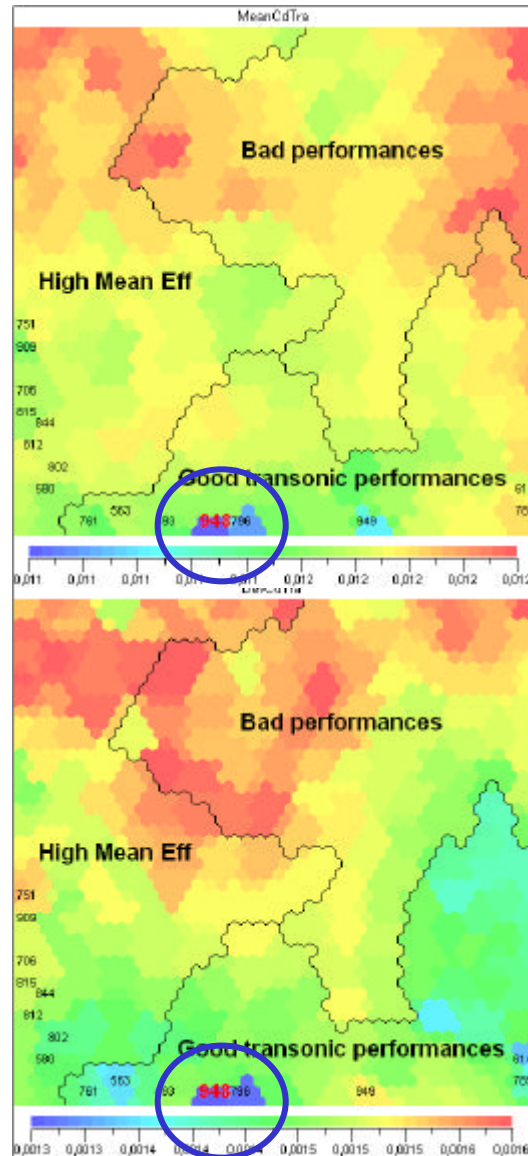
Inverse correlation between MeanEffSub and DevEffSub

No visible correlation between EffSub and CdTra

MCDM finds the best transonic performances

Visualization in Multi-D (SOM)

Correlations between design variables and performances



Correlation between V13 and transonic performances

Conclusion

- **Robust Design is highly innovative since it considers the stability of one solution;**
- **When a Robust Design Optimization is needed, the best way is a Multi Objective Algorithm (Game Theory: Nash approach);**
- **The use of SOM is very interesting for the visualization in Multi-D space (Multi Objective, multi variables)**